
 OSTİM TEKNİK ÜNİVERSİTESİ A N K A R A	FACULTY OF ENGINEERING AERO 310 COURSE SYLLABUS	Doküman No	MF.FR.003
		Revizyon Tarihi	13.11.2024
		Revizyon No	01
		Sayfa No	1 / 4

AERO 310 – Flight Mechanics				
Course Code	Course Name		Semester	
AERO 310	Flight Mechanics		Fall <input type="checkbox"/> Spring <input checked="" type="checkbox"/> Summer <input type="checkbox"/>	
Hours			Credit	ECTS
Theory	Practice	Lab	3	6
3	0	0		


Course Details	
Department	Aerospace Engineering
Course Language	English
Course Level	Undergraduate <input checked="" type="checkbox"/> Graduate <input type="checkbox"/>
Mode of Delivery	Face to Face <input checked="" type="checkbox"/> Online <input type="checkbox"/> Hybrid <input type="checkbox"/>
Course Type	Compulsory <input checked="" type="checkbox"/> Elective <input type="checkbox"/>
Course Objectives	<p>Aircraft geometry from stability and control point of view. Control surfaces.</p> <p>Governing equations related to longitudinal and lateral static stability and control of an aircraft.</p> <p>Aircraft trim analysis. Importances of the neutral point, maneuver point etc. Differences of stick fixed and stick free stability.</p> <p>Governing equations of motion for 6 degree of freedom dynamic motion of aircraft.</p> <p>Effects of the aerodynamic design, center of gravity location and moments of inertia on static and dynamic stability and control of an aircraft.</p> <p>Calculation of the stability derivatives of an aircraft.</p> <p>Dynamic modes of an aircraft.</p> <p>Solution of related problems.</p>
Course Content	<p>Aircraft Forces and Subsystems: The atmosphere, aerodynamic forces, propulsion subsystem. Turbojets-Level</p> <p>Flight in the Vertical Plane: Governing equations, level flight, ceiling, cruise flight and range, maximum endurance. Other</p> <p>Flights in the Vertical Plane: Take-off and landing, climbing flight, unpowered flight. Turning Flight in the Horizontal</p> <p>Plane: Governing equations, maximum load factor, bank angle, turning rate, and turning radius. Piston-Props-Level</p>

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	<p>Flight in the Vertical Plane: Governing equations, level flight and ceiling conditions, best range, maximum endurance.</p> <p>Other Flight: Take-off and landing, climbing flight, turning fight, turboprops, turbofans, and others-turboprops and turbofans, Mach number representation, flight and maneuvering envelops, the effect of wind on performance.</p>
Course Method/ Techniques	Lecture <input checked="" type="checkbox"/> Question & Answer <input type="checkbox"/> Presentation <input checked="" type="checkbox"/> Discussion <input type="checkbox"/>
Prerequisites/ Corequisites	
Work Placement(s)	
Textbook/References/Materials	
<ul style="list-style-type: none"> Airplane Flight Dynamics and Automatic Flight Controls, Jan Roskam Aircraft Dynamics: From Modeling to Simulation, M. R. Napolitano, Wiley ETKIN, B., DUFF REID, L., "Dynamics of Flight - Stability and Control", Third Eddition, John Wiley and Sons, 1996, ISBN: 0-471-03418-5. (TL570.E75 1995) The following books are the excellent sources of reference. MCCORMICK, B. W., "Aerodynamics, Aeronautics and Flight Mechanics", Second Edition, John Wiley and Sons, 1995, ISBN: 0-471-57506-2. (TL570.M38 1995) ANDERSON, J. D., "Introduction to Flight", Third Edition, McGraw-Hill, 1989, ISBN: 0-07-100496-3. (TL570.A68 1989) 	

Course Category				
Mathematics and Basic Sciences	<input type="checkbox"/>		Education	<input type="checkbox"/>
Engineering	<input checked="" type="checkbox"/>		Science	<input type="checkbox"/>
Engineering Design	<input type="checkbox"/>		Health	<input type="checkbox"/>
Social Sciences	<input type="checkbox"/>		Profession	<input type="checkbox"/>

Weekly Schedule		
No	Topics	Materials/Notes
1	Introduction to Aircraft Geometry and Control Surfaces	
2	Equations of Motion and Reference Frames	
3	Review of Fundamentals of Aerodynamics	
4	Review of Fundamentals of Aerodynamics	
5	Aerodynamic and Propulsive Forces and Equations	
6	Stability and Control in Steady Flight. Trim Problem.	
7	Stability and Control in Steady Flight. Trim Problem.	
8	Midterm Exam	
9	Stability and Control in Disturbed Flight Conditions	
10	Stability and Control in Disturbed Flight Conditions	
11	Longitudinal Dynamic Stability and Response to Inputs	
12	Longitudinal Dynamic Stability and Response to Inputs	

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13	Lateral Dynamic Stability and Response to Inputs	
14	Lateral Dynamic Stability and Response to Inputs	
15	Lateral Dynamic Stability and Response to Inputs	
16	Final Exam	

Assessment Methods and Criteria		
In-term studies	Quantity	Percentage
Attendance		
Lab		
Practice		
Fieldwork		
Course-specific internship		
Quiz/Studio/Criticize		
Homework		
Presentation / Seminar		
Project		
Report		
Seminar		
Midterm Exam		
Final Exam		
	Total	100%
Contribution of Midterm Studies to Success Grade		
Contribution of End of Semester Studies to Success Grade		
	Total	100%

ECTS Allocated Based on Student Workload			
Activities	Quantity	Duration (Hrs)	Total Workload
Course Hours	14	3	48
Lab			
Practice			
Fieldwork			
Course-specific Work Placement			
Out-of-class study time	14	4	56
Quiz/Studio/Criticize			
Homework	3	15	45
Presentation / Seminar			
Project			
Report			
Midterm Exam and Preparation for Midterm	1	2	2
Final Exam and Preparation for Final Exam	1	2	2
Total Workload			147
Total Workload / 25			
ECTS Credit			5

Course Learning Outcomes

No	Outcome
L1	Explains the fundamental principles of flight mechanics and aerodynamic forces.
L2	Calculates aircraft performance parameters (range, climb, turn, etc.).
L3	Analyzes aircraft stability and control characteristics.
L4	Derives and simulates flight dynamics equations.
L5	Evaluates flight safety and certification processes.

Contribution of Course Learning Outcomes to Program Competencies/Outcomes

Contribution Level: 1: Very Slight, 2: Slight, 3: Moderate, 4: Significant, 5: Very Significant

	P1	P2	P3	P4	P5	P6	P7	P8	P9	P10	P11	Total
L1	5	4	3	3	2	2	2	3	3	2	2	31/55 = %56.36
L2	5	5	4	4	3	2	2	3	3	2	3	36/55 = %65.45
L3	4	5	5	4	4	3	3	3	3	3	3	40/55 = %72.73
L4	4	4	4	5	5	3	3	3	3	3	3	40/55 = %72.73
L5	3	3	3	3	3	4	4	4	5	4	4	40/55 = %72.73
Total												187/275= 68%